

Design of Nano-Satellite Cluster Formations for Bi-Directional Reflectance Distribution Function Estimations

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***Bi-Directional
Reflectance Function
(BRDF) Estimations***

falls under the broad topic
of

**Multi-Angular, Multi-
Spectral Remote
Sensing of the Earth**

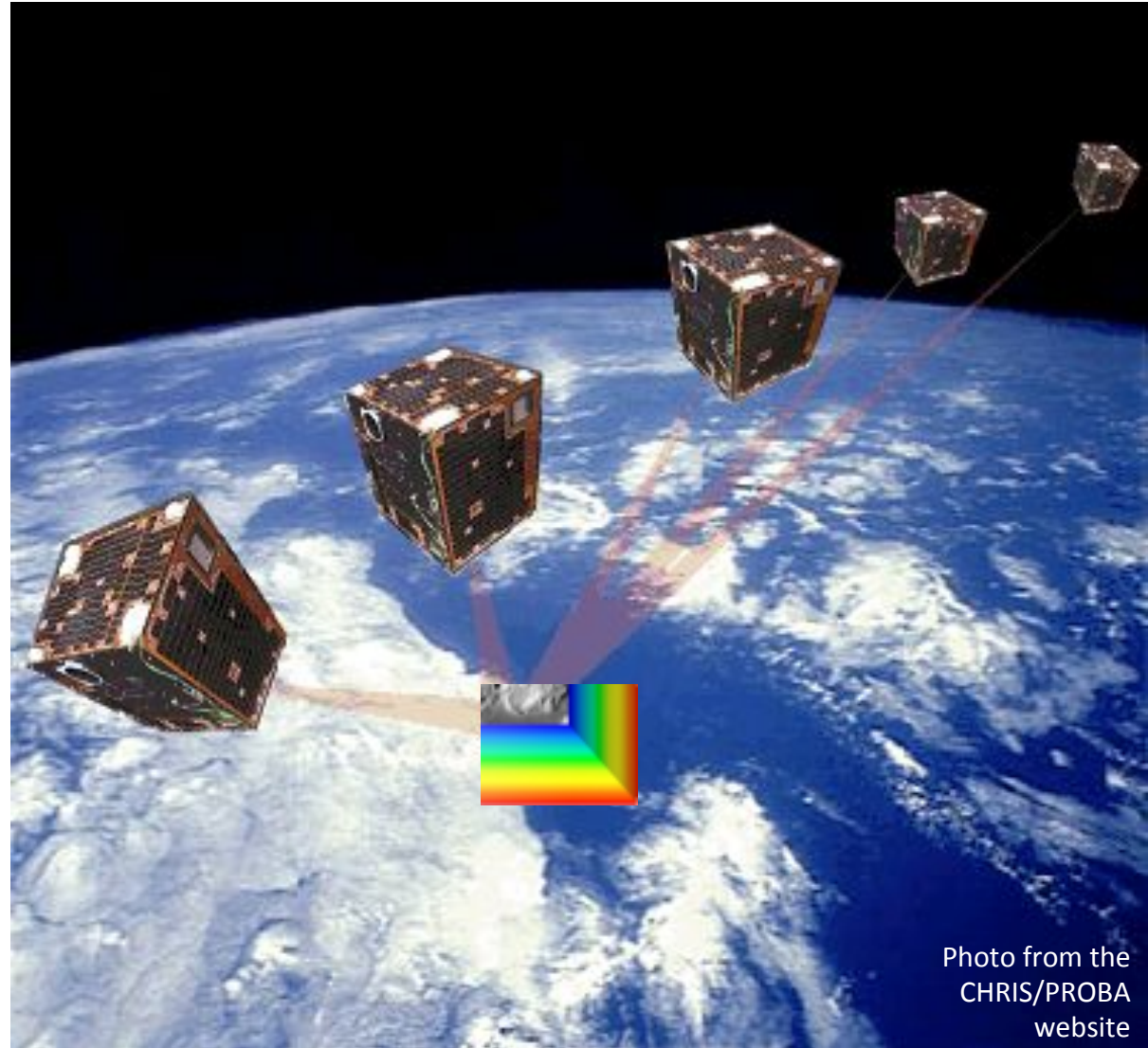
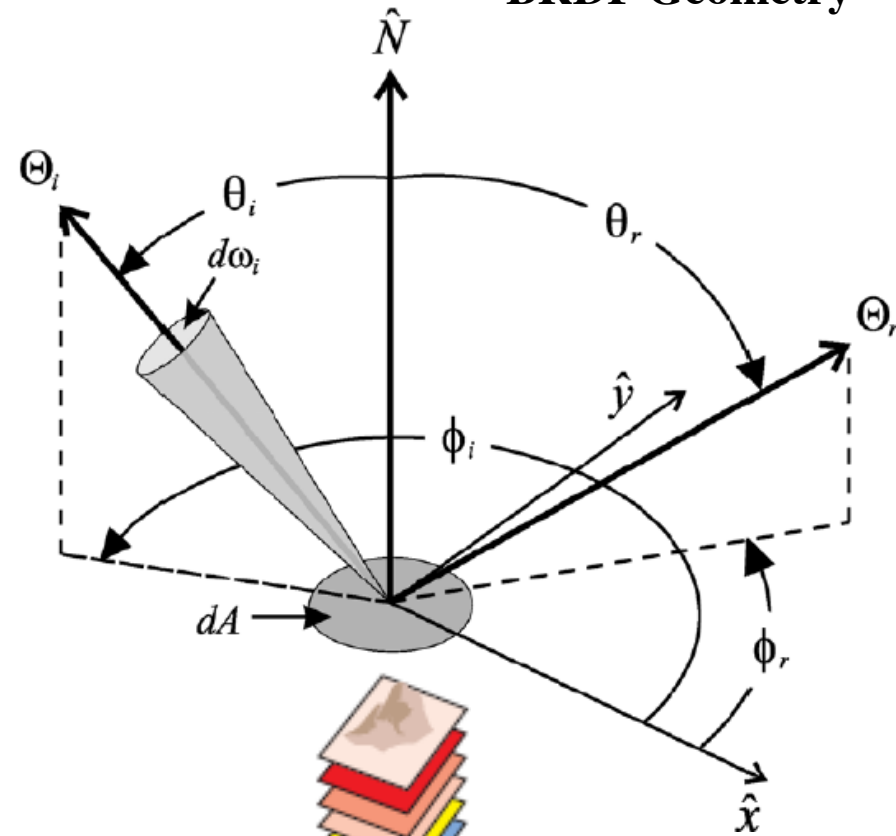


Photo from the
CHRIS/PROBA
website

BRDF Definition

- Bi-directional reflectance distribution function
- Anisotropic (angle-dependent) and multispectral (near-solar spectrum) reflectance of clouds and ground surface
- $R(\theta_i, \theta_r, \phi_i, \phi_r, \lambda)$

BRDF Geometry

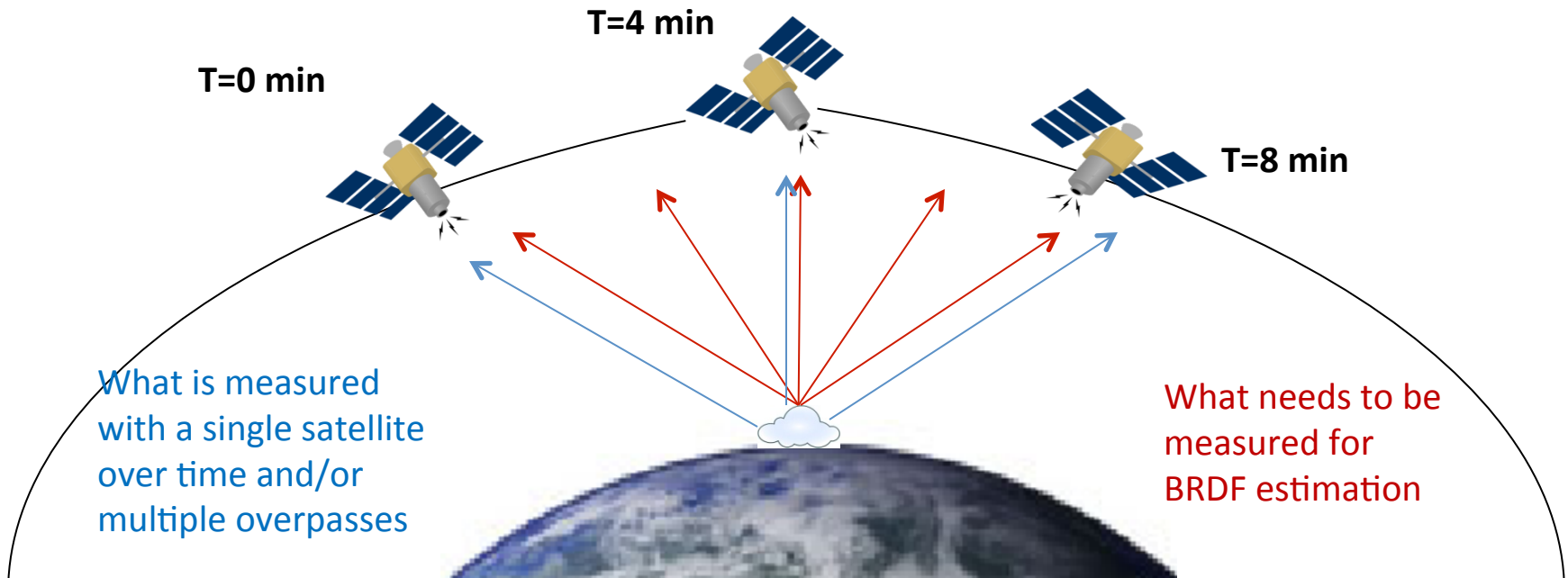


BRDF Ground Spot spectrum



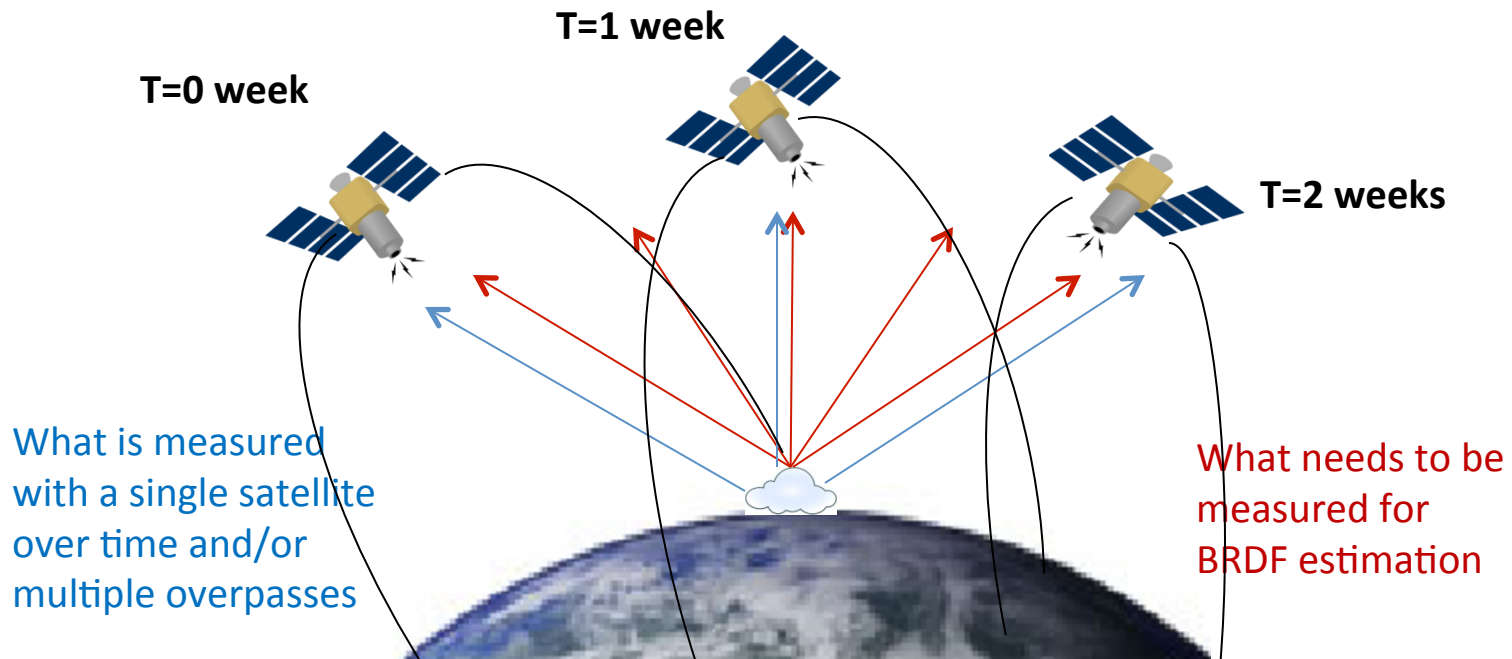
BRDF Estimation by combining the consecutive measurements

- Problem:**
1. Restrictive plane with respect to the sun
 2. Up to 10 minutes between measurements



BRDF Estimation by combining measurements over consecutive overpasses

- Problem:**
1. Restrictive plane with respect to the sun
 2. Up 2 weeks between measurements

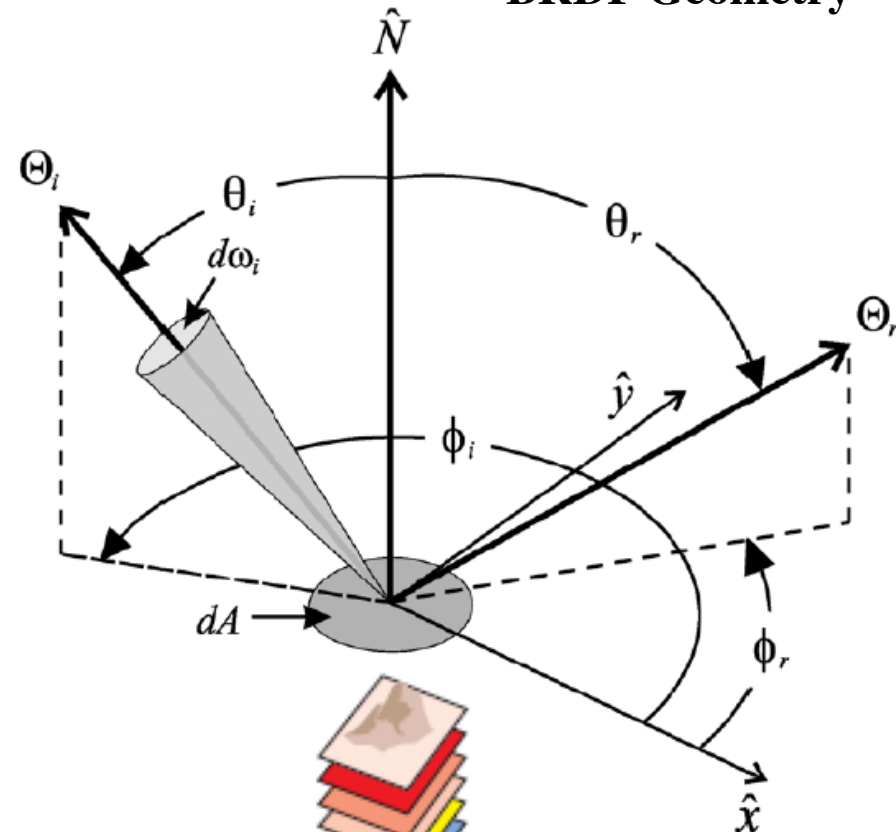


Theoretical function, very important applications

Mentioned in many science and policy docs

- "Responding to the challenge of Climate and Environmental Change"; "enhance understanding of the role of CO₂ in the global carbon cycle" – **NAS Decadal Survey 2007**
- "To provide data on variables (surface BRDF and albedo) that have wide application... (especially those) designed primarily for cloud and aerosol studies" - **Ecosystems Structure and Biomass panel on Multi Angle Remote Sensing**

BRDF Geometry



BRDF Ground Spot spectrum



BRDF effects on important applications such as **albedo** radiative forcing, gross primary productivity is stark

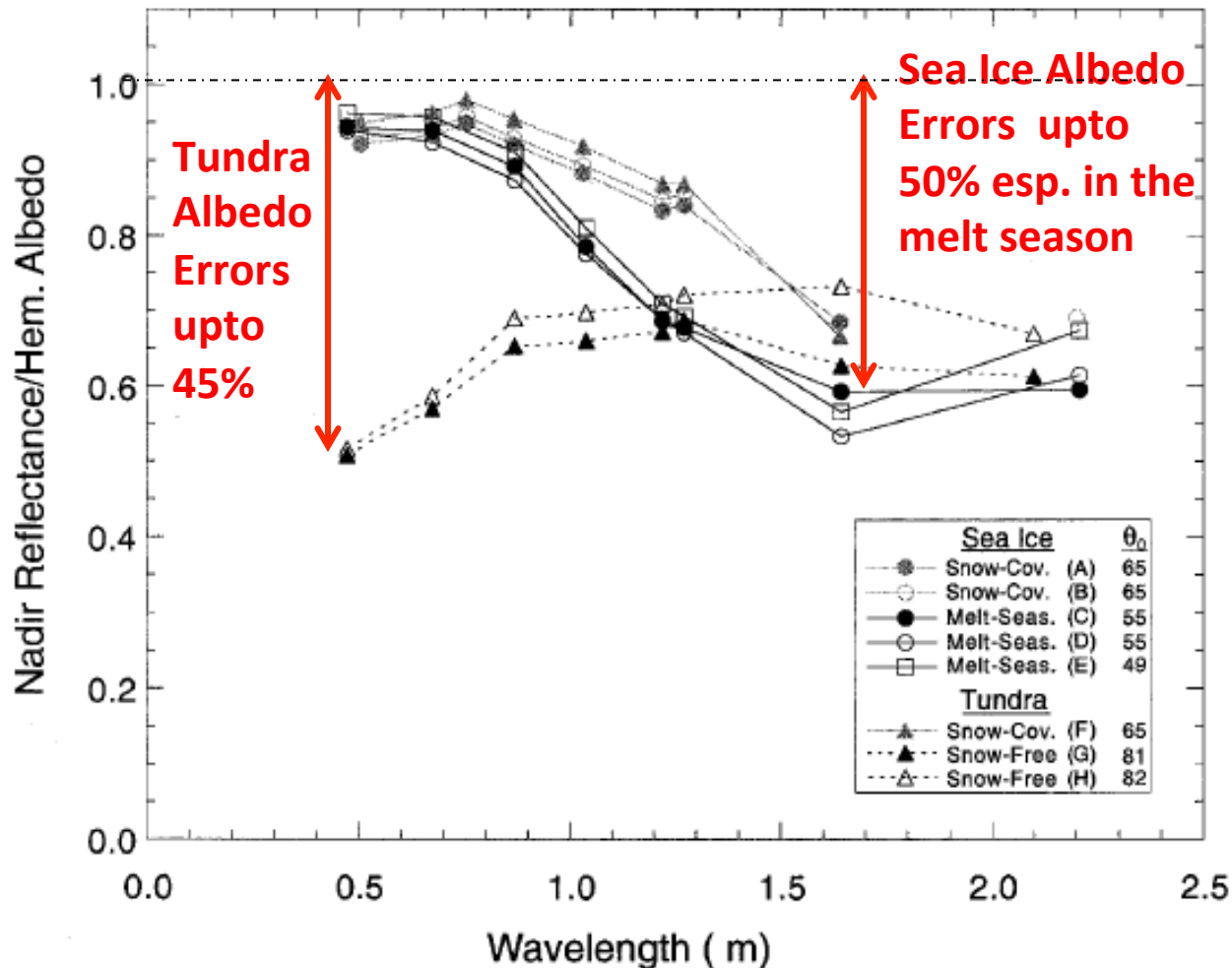
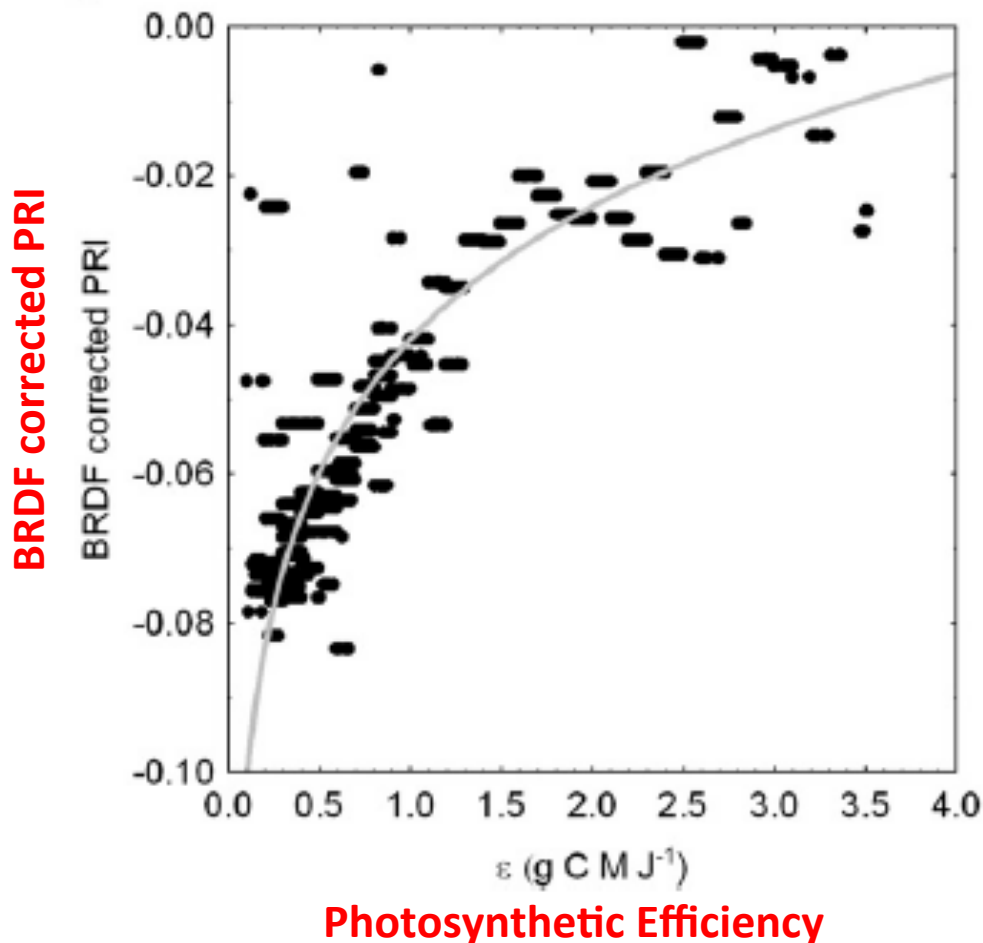


Image Credits: Arnold et. al, 2002

Figure uses thousands of angular measurement data from the airborne Cloud Absorption Radiometer taken during the ARM-CAS campaign in 1998.

BRDF effects on important applications such as albedo radiative forcing, **gross primary productivity** is stark



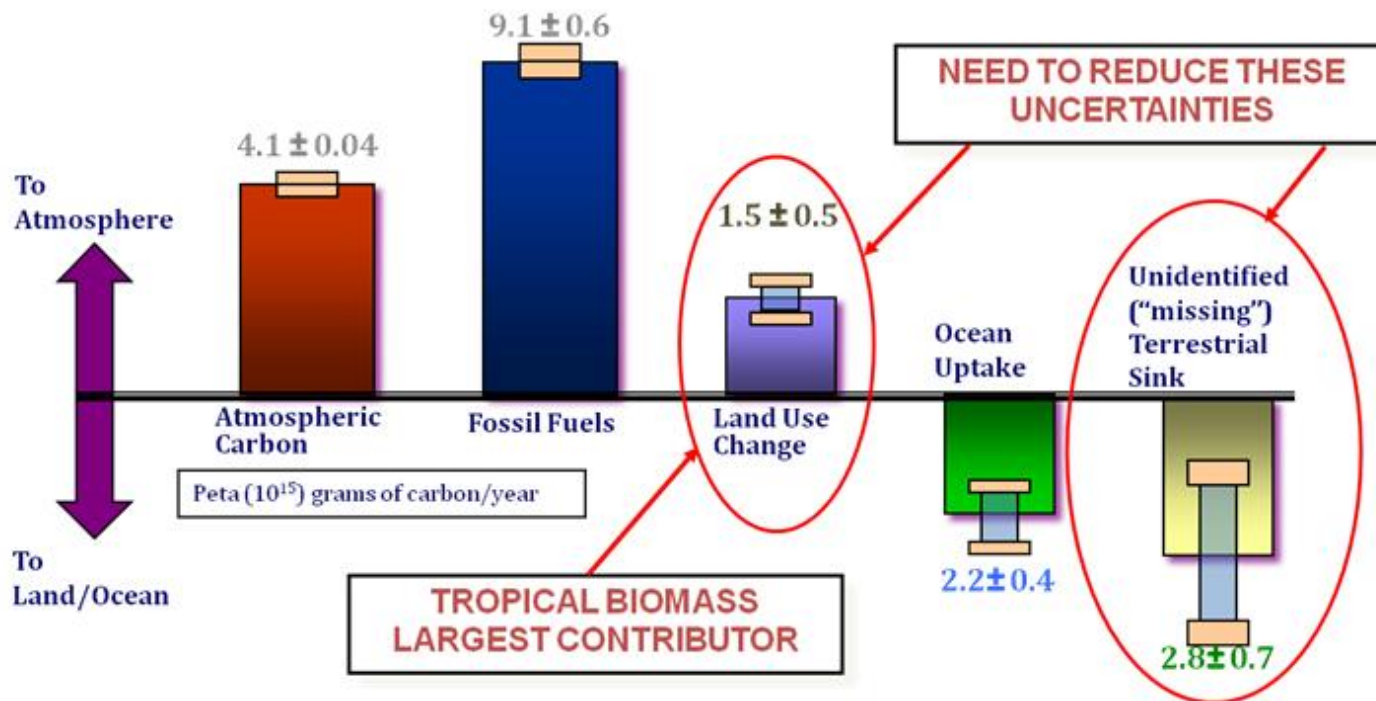
GPP => Extent to which vegetation acts as a Carbon Dioxide Sink

GPP \propto Photosynthetic Efficiency \propto BRDF corrected Photosynthetic Refractive Index

Image Credits: Hilker et. al., 2008

BRDF effects on important applications such as albedo radiative forcing, **gross primary productivity** is stark

Global Carbon Budget
PgC/Yr 2000-2006
(Canadell et al., 2007)



40% errors in current budget estimates shown in *Canadell et al. 2007*

Reduced to 10% errors using CHRIS multi-angular data shown in *Hall and Tucker 2010*

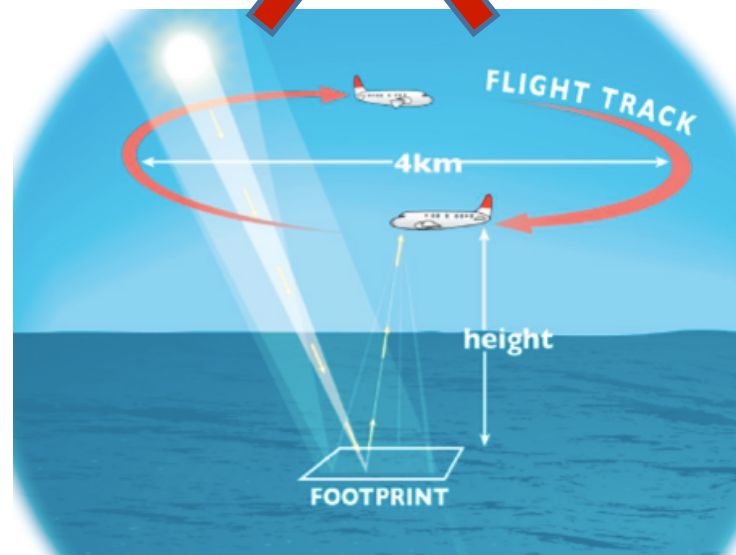
Airborne: Very accurate for local BRDF estimation
 e.g. Cloud Absorption Radiometer (CAR)

BUT no global or continuous coverage, expensive to scale up area and time

Geometrical Requirements

Spectral Requirements

BRDF-Science Metrics	Geometrical Requirements			Spectral Requirements	
	Number of angles	Ground Pixel Size in km X km	Revisit time (any view angles)	Spectral Range	# of spectral bands



Spaceborne: Angular coverage through Large swath or FOV¹,
Fwd-Aft sensors², autonomous maneuverability³

BUT fall short in terms of science metric/s + nearing EOL

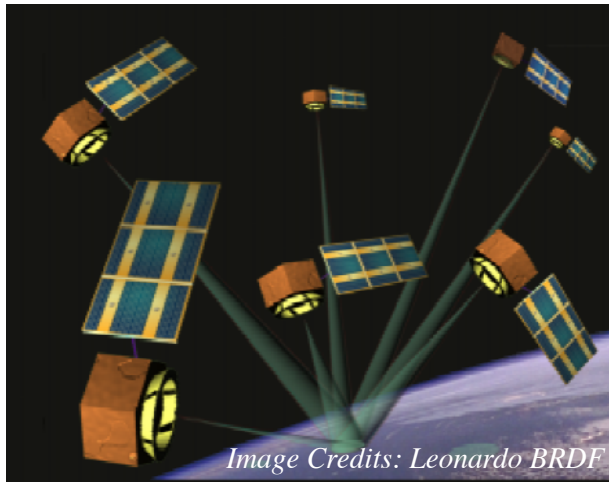
Geometrical Requirements

Spectral Requirements

BRDF-Science Metrics Current Instruments ↓	Number of angles	Ground Pixel Size in km X km	Revisit Time (any view) in days	Spectral Range	# of spectral bands
¹ MODIS	1	0.25 to 1	~2(16day RGT)	0.4-14.4 μm	36
¹ POLDER	14	6 X 7	~2(16day RGT)	0.42-0.9 μm	9
¹ CERES	1	10 to 20	~2(16day RGT)	0.3-12 μm	3
² MISR	9	0.275 to 1.1	9(16 day RGT)	0.44-0.87 μm	4
² ATSR	2	1 to 2	3-4	0.55-12 μm	7
² ASTER	2	0.015 to 0.09	~2(16day RGT)	0.52-11.65 μm	14
³ CHRIS	5-15	0.017 to 0.5	As per command	0.415-1.05 μm	18-63

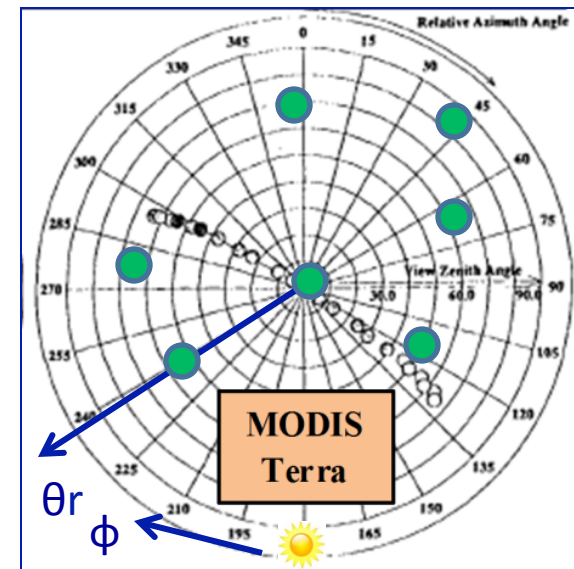
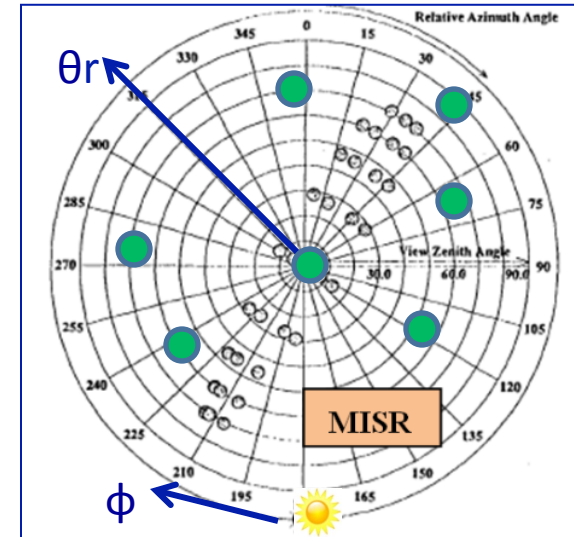
Major Gap: Angular undersampling (θ_s, θ_r, ϕ)

Potential Solution: Clusters of nano-satellites since each sat will be small

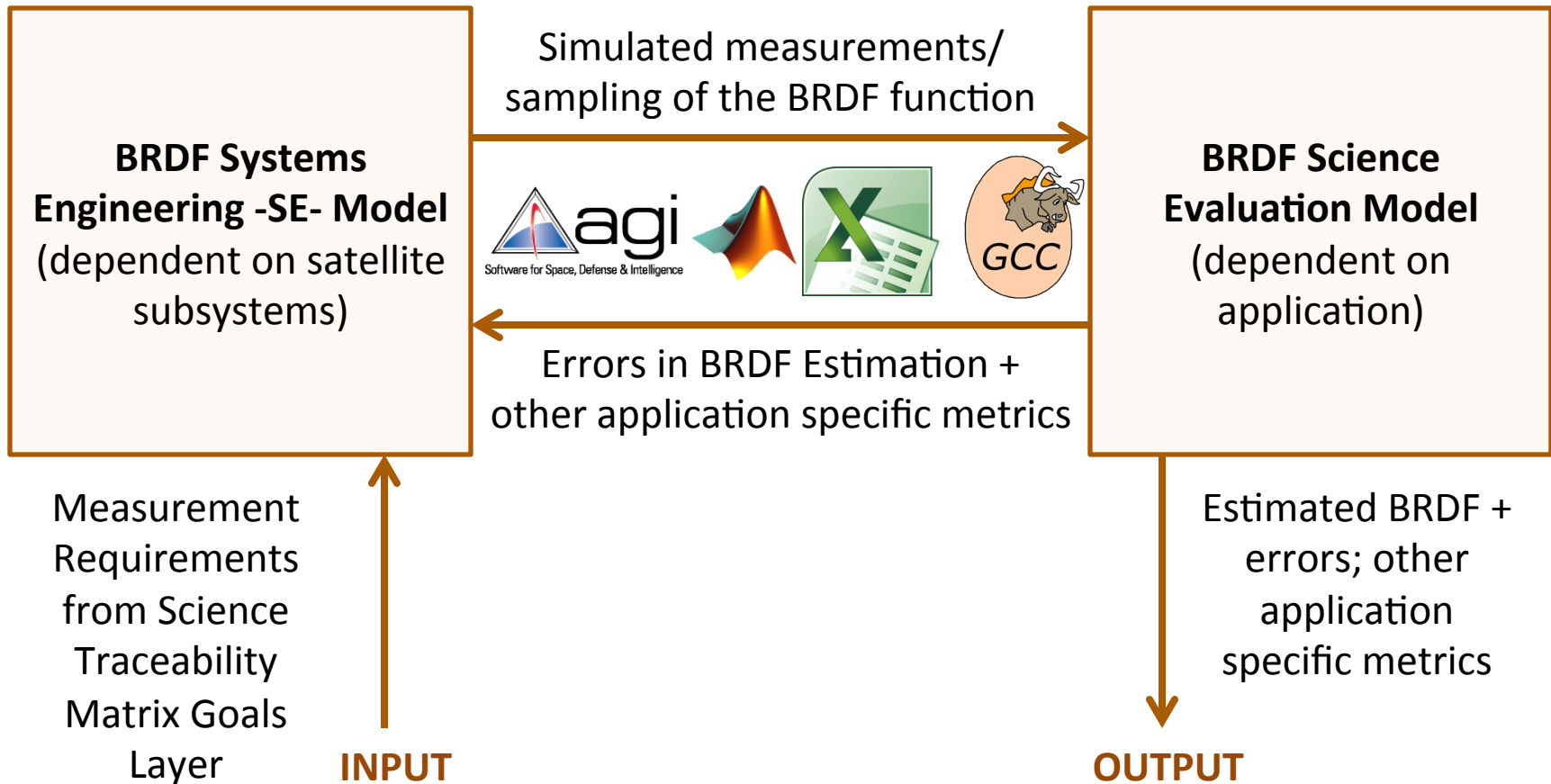


Additional advantages:- 6U cubesats under development, Standard bus, Secondary payload launches, Cubesat GS network

Disadvantages:- Restrictive h-i combinations, mass/volume constraints

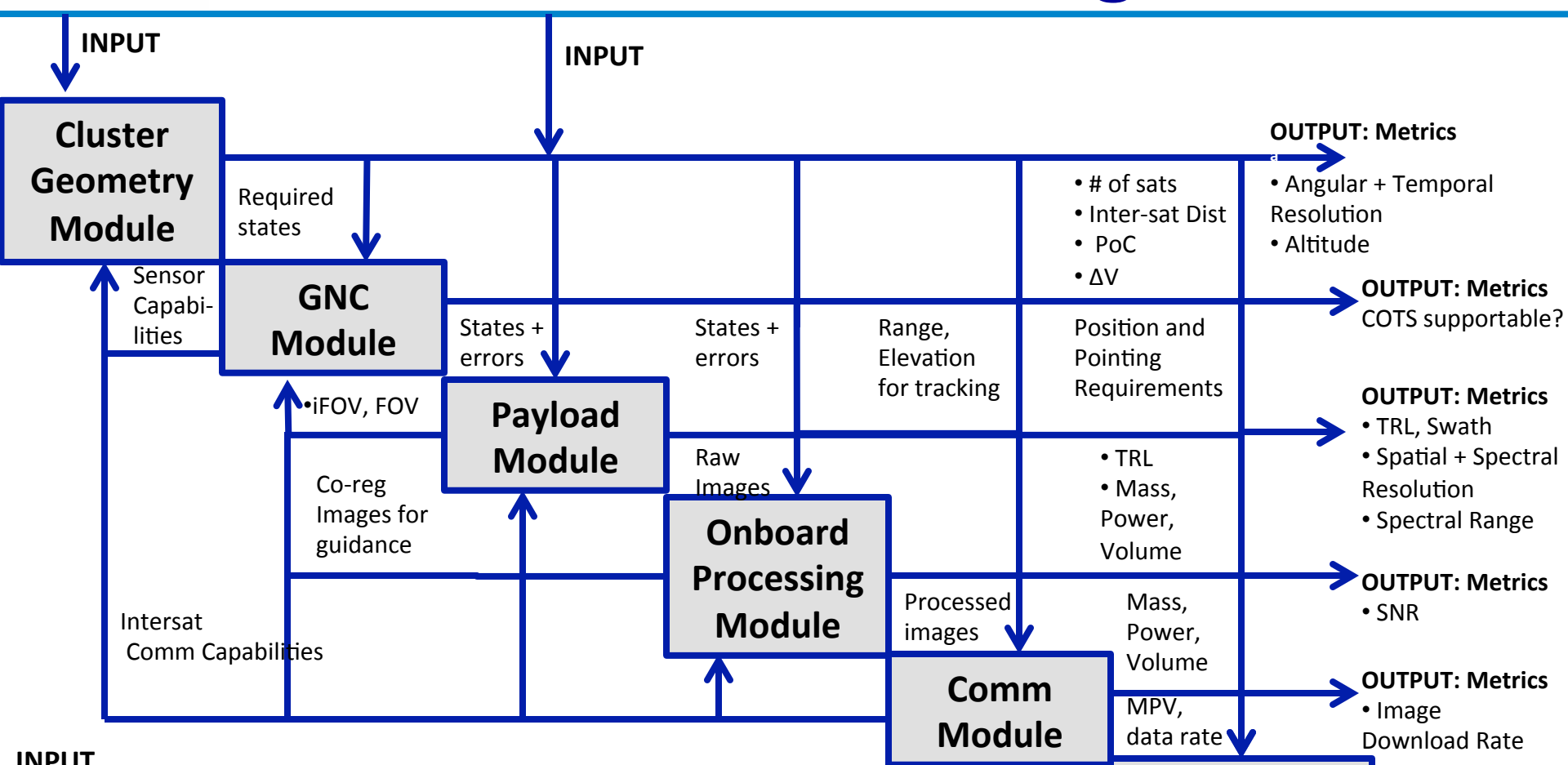


Build a Systems engineering (SE) model integrated with traditional BRDF Estimation models to finalize the ideal cluster architecture, satellite design, subsystem design and primary instrument





SE Model as an N2 Diagram



INPUT

BRDF Measurement Requirements

- Spatial resolution (< 500 m)
- Measurement Zenith Angles (< 60°)
- Measurement Azimuth (< 360°)
- Solar zenith Angles (< 80°)
- Spectral Resolution (>14 bands)
- Spectral Range (350-2300 nm)

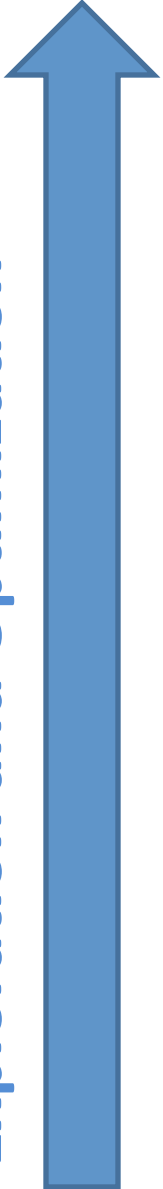
Nanosat Bus Requirements

- Mass < 10 kg
- Cube < 10X20X30 cm
- Power < 25 W
- Altitude (400 – 800 km)

OUTPUT: Metrics Architecture "Size"

- Relative Complexity
- Mass, Cost

Computational Ease of Tradespace
Exploration and Optimization



LINEARIZED HILL CLOHESSY WILTSHIRE EQUATIONS

DUAL SPIRAL EQUATIONS

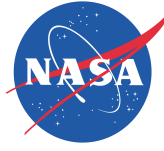
GLOBAL ORBIT PROPAGATION USING STK

Model Fidelity/Reliability





Cluster Geometry Models



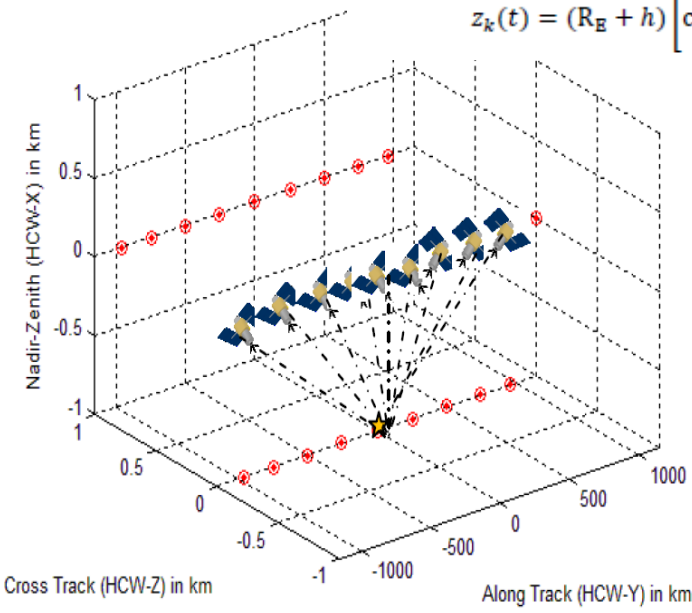
Computational Ease of Tradespace Exploration and Optimization

LINEARIZED HILL CLOHESSY WILTSHIRE EQUATIONS

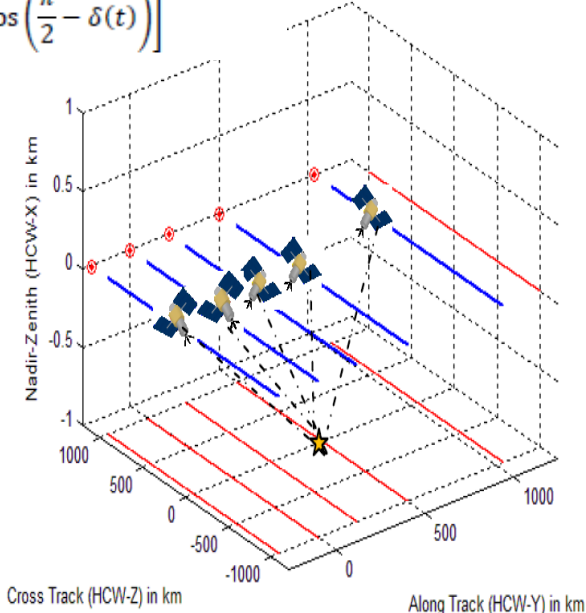
$$x_k(t) = (R_E + h) \left[\cos(\Phi_k + \alpha(t)) \sin\left(\frac{\pi}{2} - \delta(t)\right) - 1 \right]$$

$$y_k(t) = (R_E + h) \left[\sin(\Phi_k + \alpha(t)) \sin\left(\frac{\pi}{2} - \delta(t)\right) \right]$$

$$z_k(t) = (R_E + h) \left[\cos\left(\frac{\pi}{2} - \delta(t)\right) \right]$$



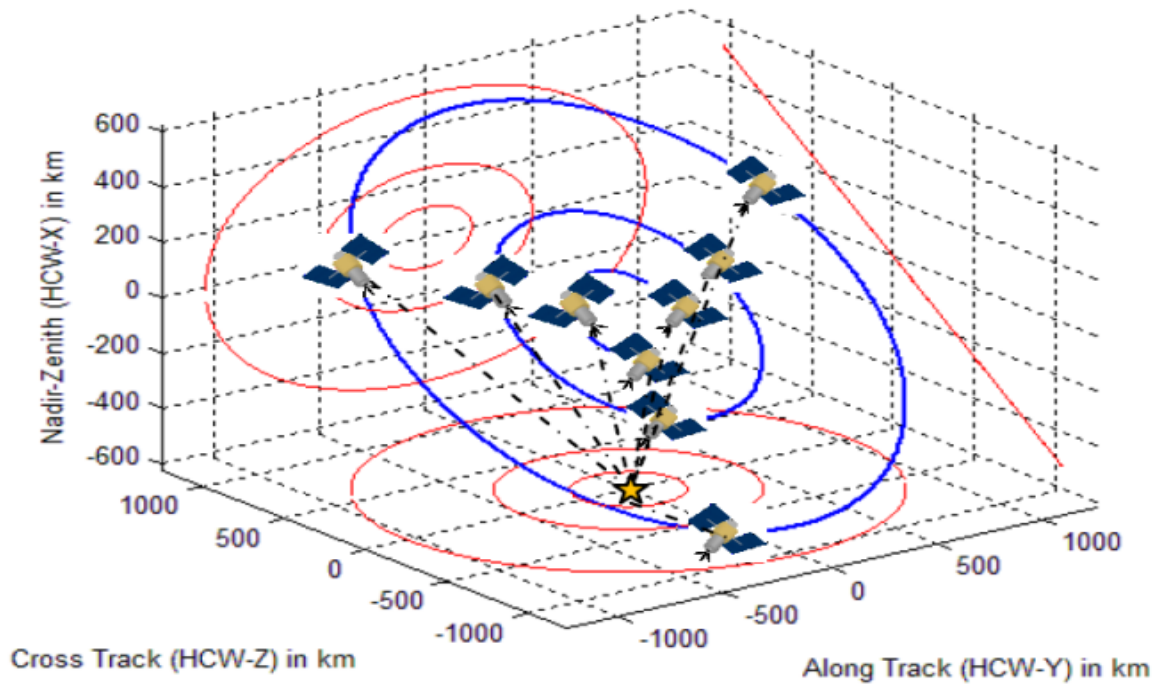
String of Pearls (SOP)



Cross Track Scan (CTS)

Model Fidelity/Reliability

LINEARIZED HILL CLOHESSY WILTSHIRE EQUATIONS



Free Orbit Ellipse (FOE)

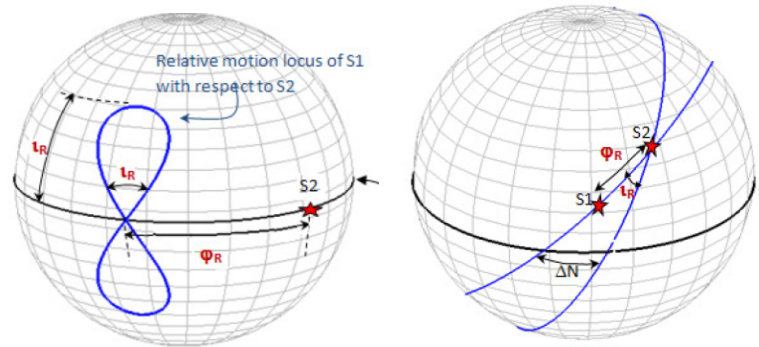
↑
Computational Ease of Tradespace
Exploration and Optimization

↓
Model Fidelity/Reliability

Computational Ease of Tradespace
 Exploration and Optimization

Model Fidelity/Reliability

*Special Case for the Dual Spiral model :
Relative Analemma*



$$\sin \delta = \sin i_R \sin nt$$

$$\alpha = nt - \text{atan}(\cos i_R \tan nt)$$

$$\Delta\Phi = \Phi_2 + \Phi_R - \Phi_1$$

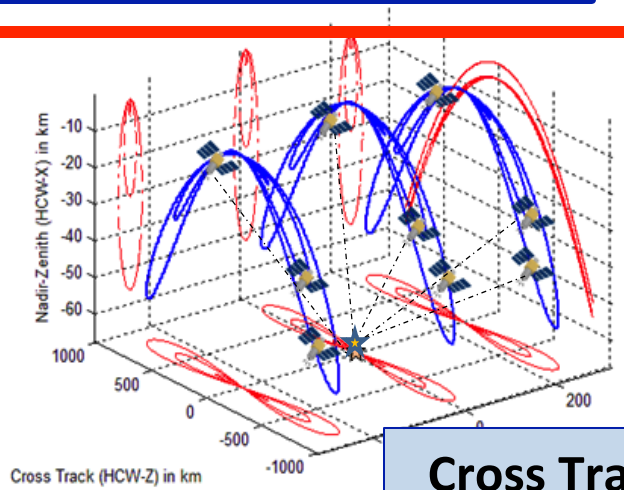
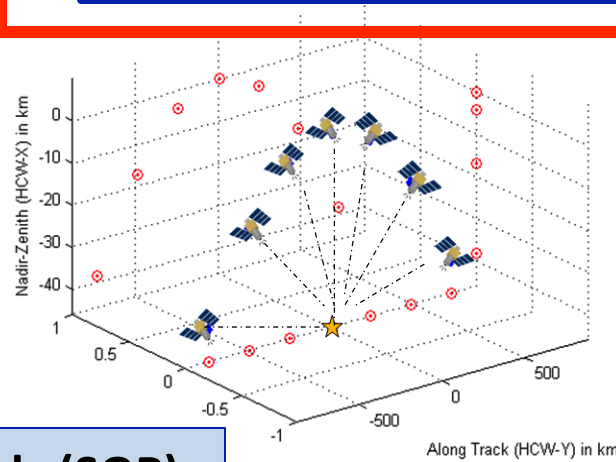
$$\cos \Phi_1 = \frac{\cos i_2 - \cos i_1 \cos i_R}{\sin i_R \sin i_2}$$

$$\cos \Phi_1 = \frac{\cos i_2 - \cos i_1 \cos i_R}{\sin i_R \sin i_2}$$

$$\cos i_R = \cos i_1 \cos i_2 + \sin i_1 \sin i_2 \cos \Delta N$$

$$\Phi_R = (T_2 - T_1)n + \Delta\Phi$$

DUAL SPIRAL EQUATIONS

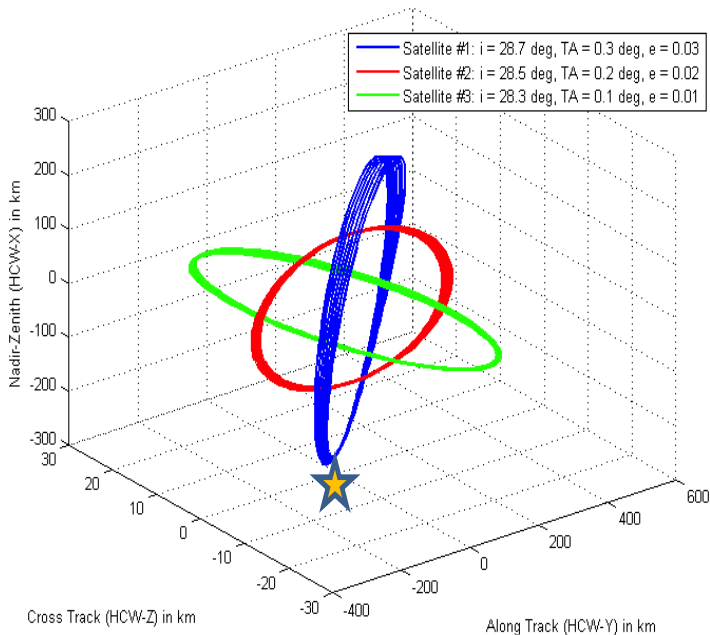


String of Pearls (SOP)

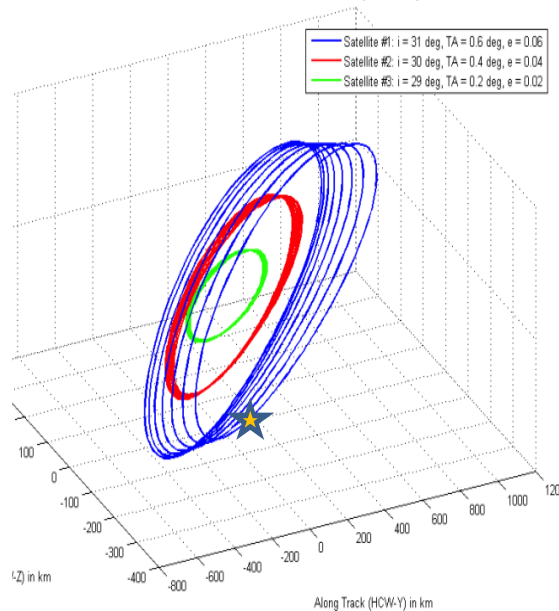
Cross Track Scan (CTS)

Free Orbit Ellipse(FOE)

Relative Motion of two satellites wrt a reference satellite at $i = 28.5$ deg, $TA = 0$ deg, $e = 0$



Relative Motion of two satellites wrt a reference satellite at $i = 28.5$ deg, $TA = 0$ deg, $e = 0$



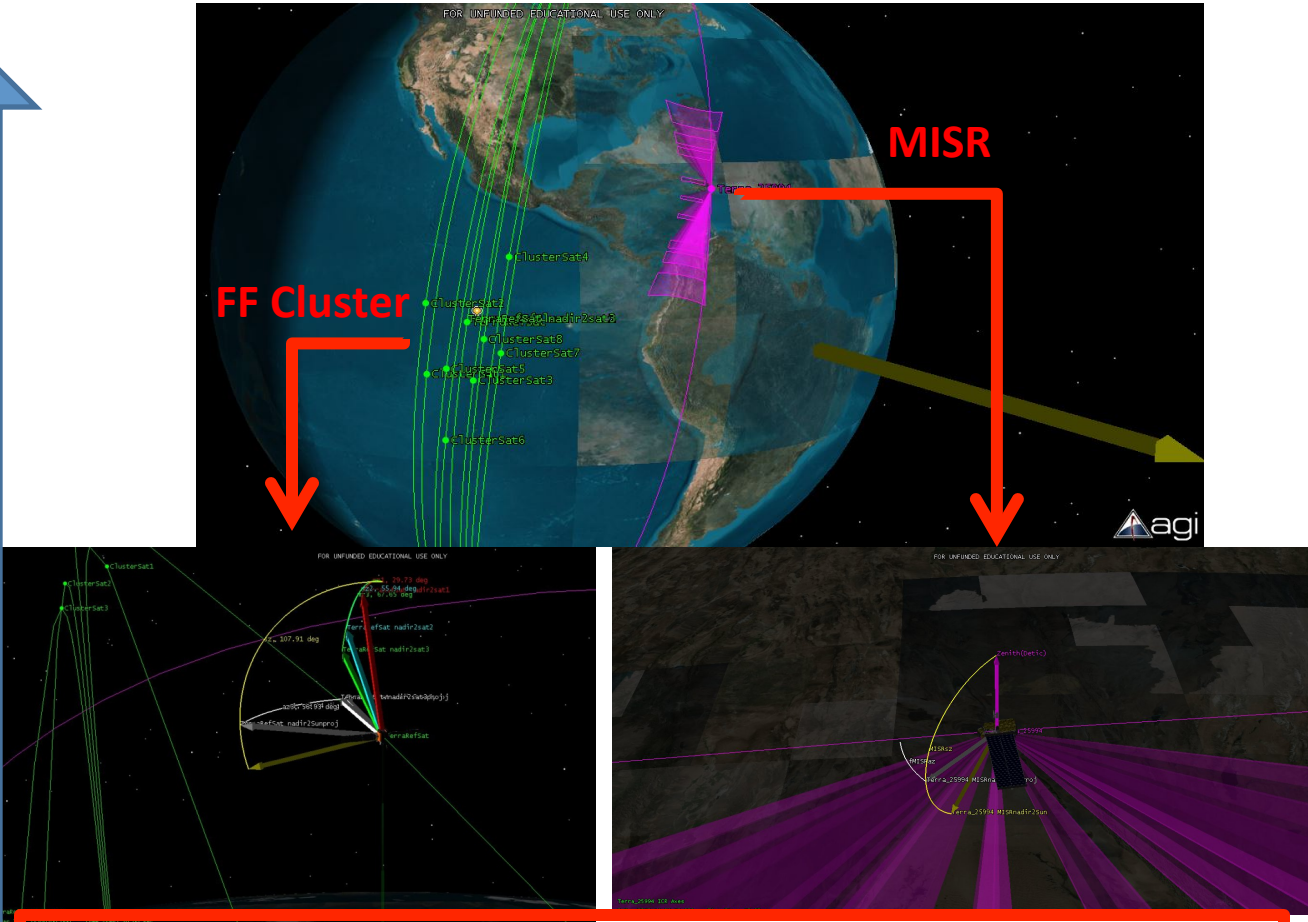
GLOBAL ORBIT PROPAGATION USING STK

Trajectories in the LVLH frame of orbits propagated for one day

Computational Ease of Tradespace
Exploration and Optimization

Model Fidelity/Reliability

Computational Ease of Tradespace
Exploration and Optimization

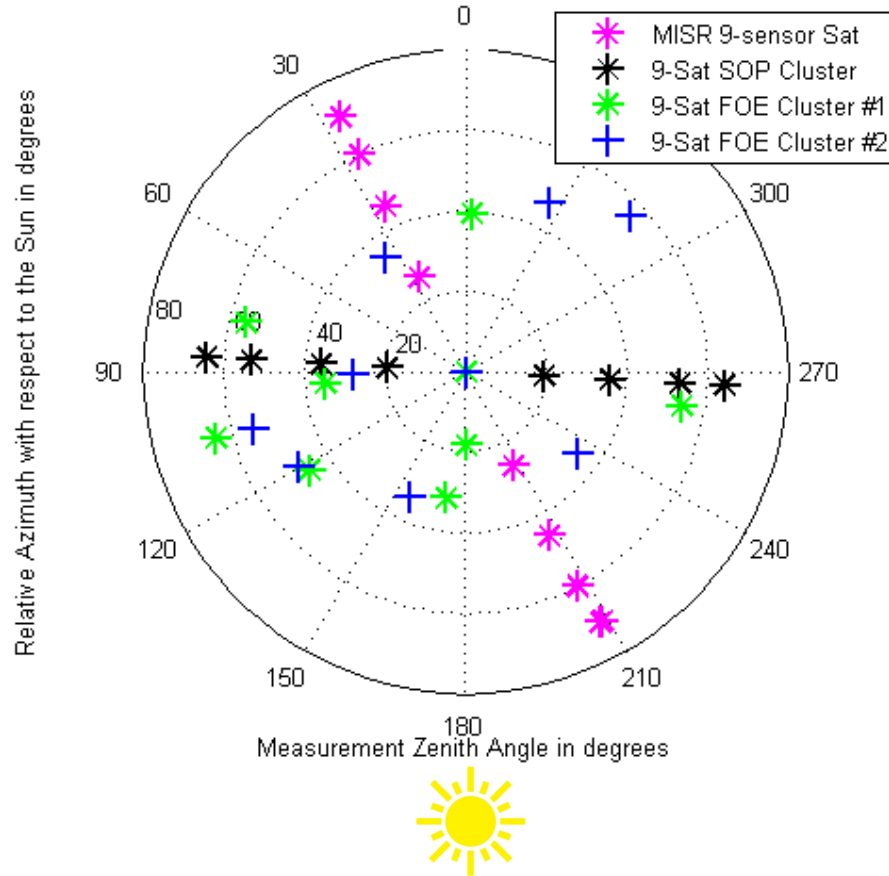


Model Fidelity/Reliability

GLOBAL ORBIT PROPAGATION USING STK

Target (yellow) imaged at multiple angles by cluster (green), MISR (pink)

Computational Ease of Tradespace
Exploration and Optimization

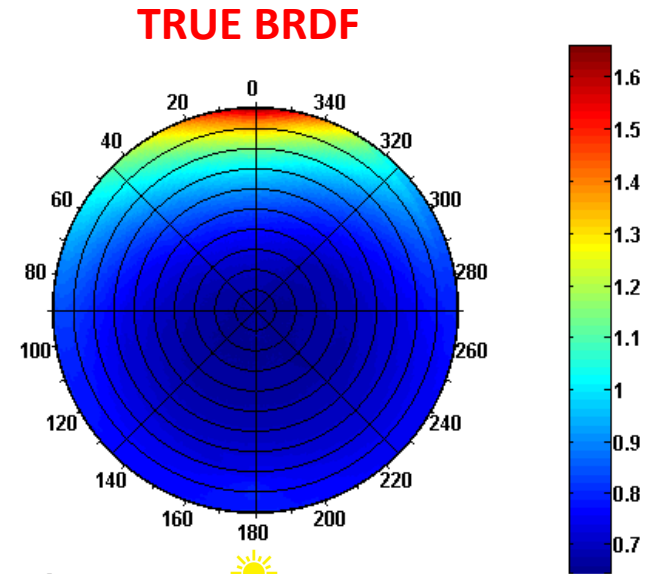


Model Fidelity/Reliability

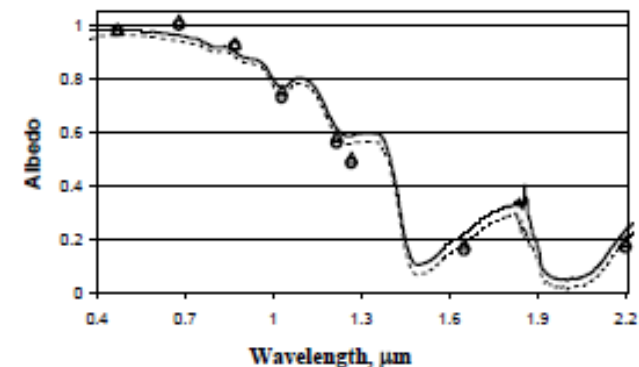
GLOBAL ORBIT PROPAGATION USING STK

Target (yellow) imaged at multiple angles by cluster (green), MISR (pink)

- ***Snow environment*** selected because of lack of aerosol effects, less clouds, important for climate change, melt season needs >1 day temporal repeat, availability of campaign.
- Used ***BRDF data as “truth”*** from the ARCTAS (Arctic) campaign at Elson Lagoon (71.3 N, 156.4 W), Alaska, which was studied by the NASA P-3B carrying the Cloud Absorption Radiometer (CAR).
- Data available at all 360 azimuth and 90 zenith angles, so ***easy to select any angular sampling combination*** for trades.
- Used the ***Rossthick-LiSparse (RLTS) model*** because it is linear, suited for spatial scales, used for MODIS products and tested appropriate for snow [*Lyapustin et al, 2010*]



RADIUS: View Zenith Angle in degrees
 AZIMUTH: View Azimuth Angle in degrees
 Wavelength = 1.02 microns (atm window)
 Acquisition height = 1.69 km



Black Sky Albedo at solar zenith angle = 30.72 deg

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- Used **BRDF data as “truth”** from the ARCTAS (Arctic) campaign at Elson Lagoon (71.3 N, 156.4 W), Alaska, which was studied by the NASA P-3B carrying the Cloud Absorption Radiometer (CAR).
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$$BRDF(\theta_s, \theta_v, \Delta\phi, \lambda) \cong R(\theta_s, \theta_v, \Delta\phi, \Lambda)$$

$$= f_{iso}(\Lambda) + f_{vol}(\Lambda)K_{vol}(\theta_s, \theta_v, \Delta\phi) + f_{geo}(\Lambda)K_{geo}(\theta_s, \theta_v, \Delta\phi, P_4, P_5)$$

where:

$$K_{vol} = \frac{(\pi/2 - \xi) \cos\xi + \sin\xi}{\cos\theta_s + \cos\theta_v} - \frac{\pi}{4}$$

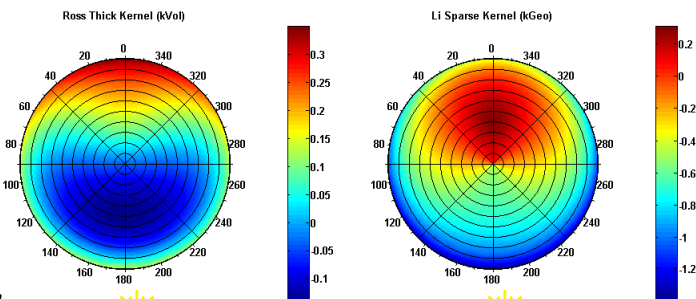
$$\cos\xi = \cos\theta_s \cos\theta_v + \sin\theta_s \sin\theta_v \cos\Delta\phi$$

$$K_{geo} = \frac{1 + \sec\theta'_s \sec\theta'_v + \tan\theta'_s \cos\Delta\phi}{2} + \left[\frac{t - \sin t \cos t}{\pi} - 1 \right] (\sec\theta'_s + \sec\theta'_v)$$

$$\cos^2 t = \min \left\{ \left[\frac{P_4}{\sec\theta'_v + \sec\theta'_s} \right]^2 [D^2 + (\tan\theta'_v \tan\theta'_s \sin\Delta\phi)^2], 1 \right\}$$

$$\tan\theta'_x = P_5 \tan\theta_x; \quad x = v \text{ or } s$$

$$D = \sqrt{\tan^2\theta'_s + \tan^2\theta'_v - 2 \tan^2\theta'_s \tan\theta'_v \cos\Delta\phi}$$



RADIUS: View Zenith Angle in degrees
 AZIMUTH: View Azimuth Angle in degrees
 SZA = 30.72 deg



Snow Albedo Application



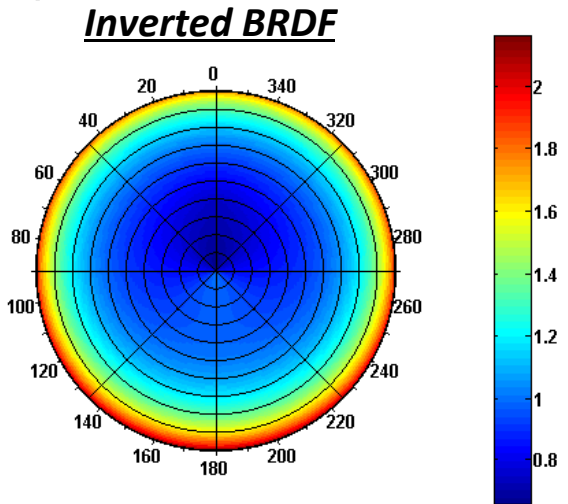
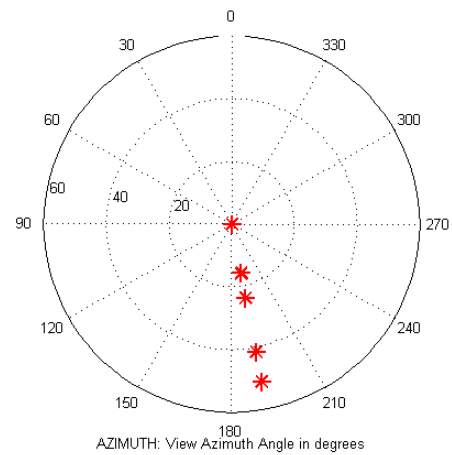
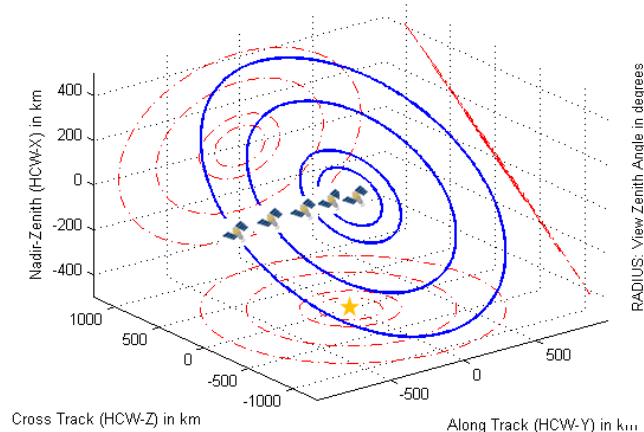
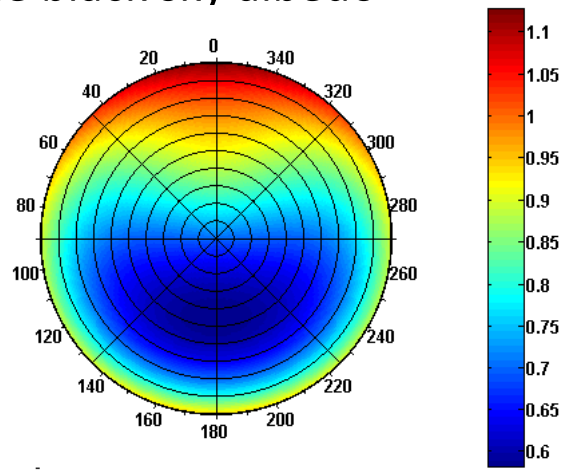
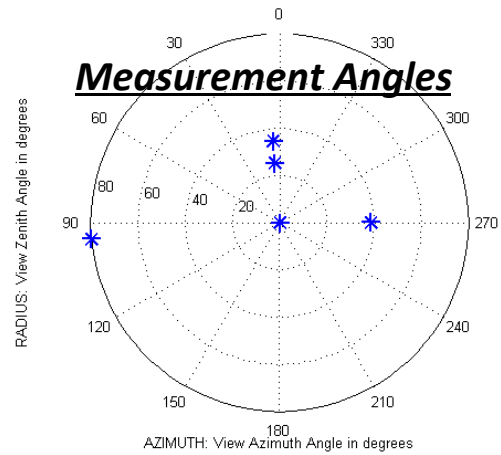
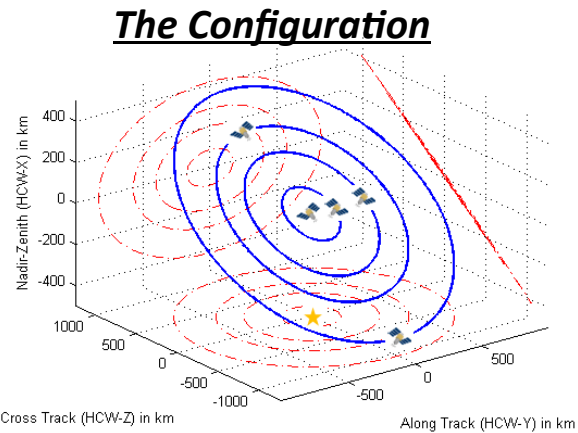
KEY TRADE VARIABLES: Cluster Geometry (HCW-FOE), FOE Orientation (22deg to LVLH-XY), # Rings (2-4), # Satellites (**5, 9, 13**), Angle subtended by rings at chief orbit (20-60deg), Orbit Orientation (normalized to Sun), Angular Coverage

INTEGRATED MODEL: Modified HCW for FF + RLTS for BRDF Estimation

KEY METRICS: RMS error wrt true BRDF, Albedo error wrt true black sky albedo

Best Case
RMS err=0.08
Albedo err=0.0015

Worst Case
RMS err=0.3892
Albedo err=0.2928





Snow Albedo Application



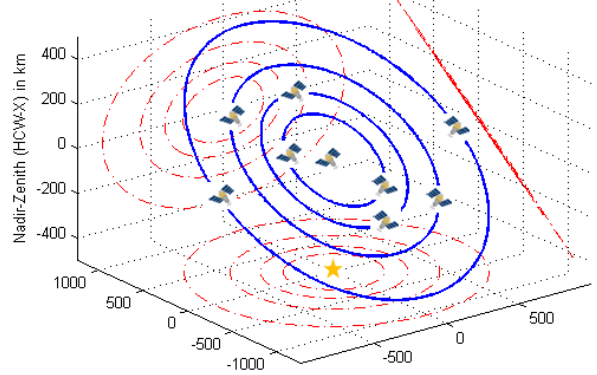
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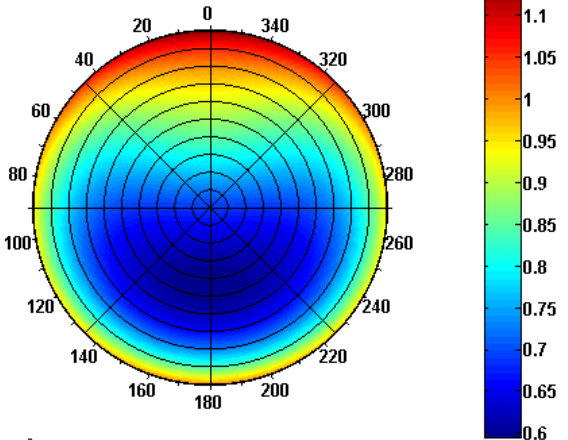
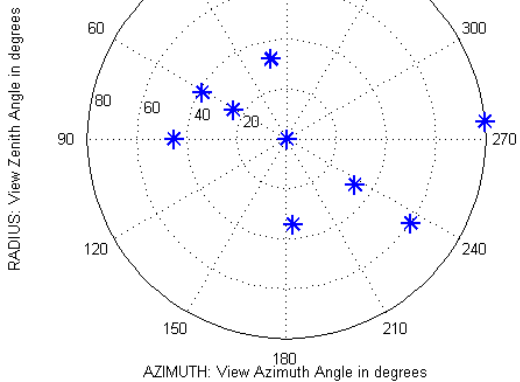
KEY METRICS: RMS error wrt true BRDF, Albedo error wrt true black sky albedo

Best Case
RMS err=0.0707
Albedo err=0.001

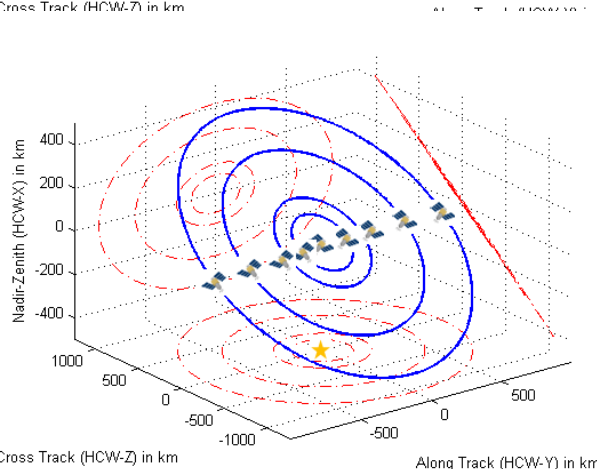
The Configuration



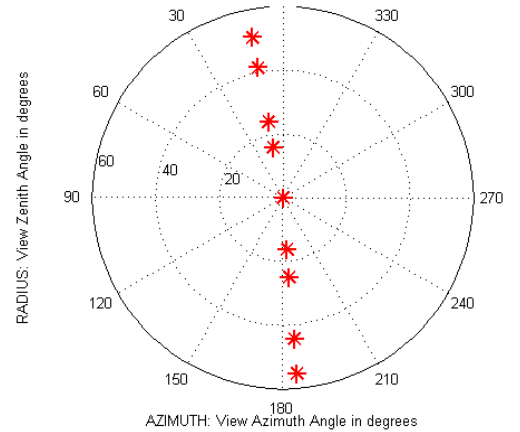
Measurement Angles



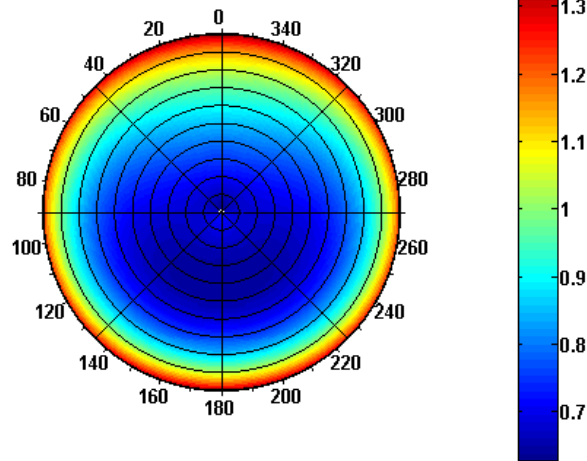
Worst Case
RMS err=0.1137
Albedo err=0.045



Measurement Angles



Inverted BRDF





Snow Albedo Application



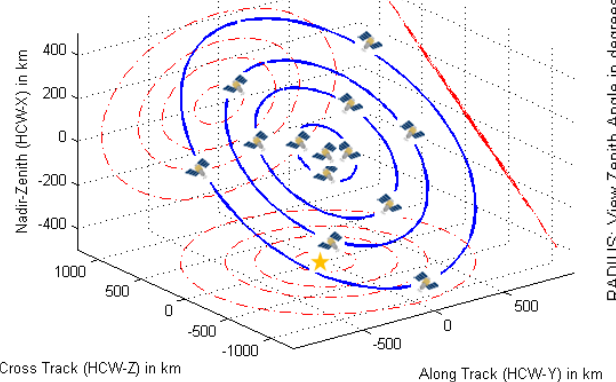
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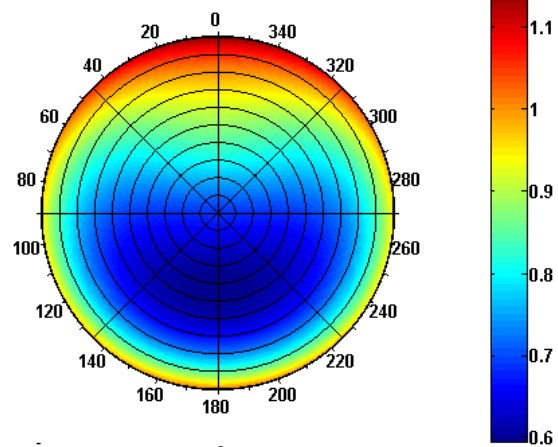
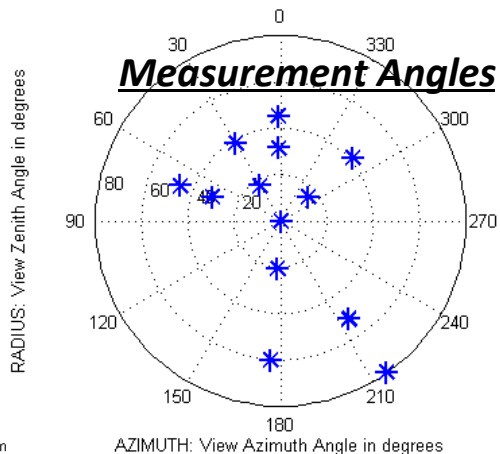
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Best Case
RMS err=0.07
Albedo err=0.0004

The Configuration

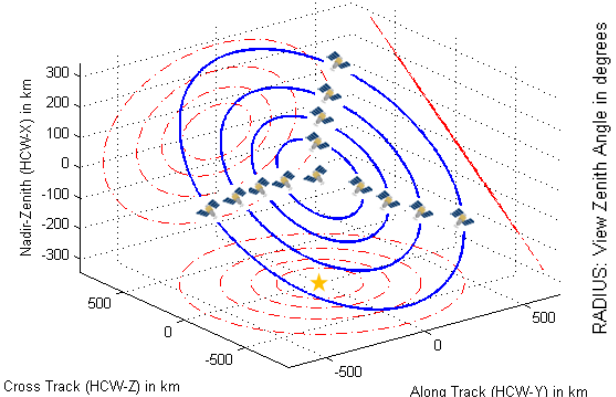


Measurement Angles

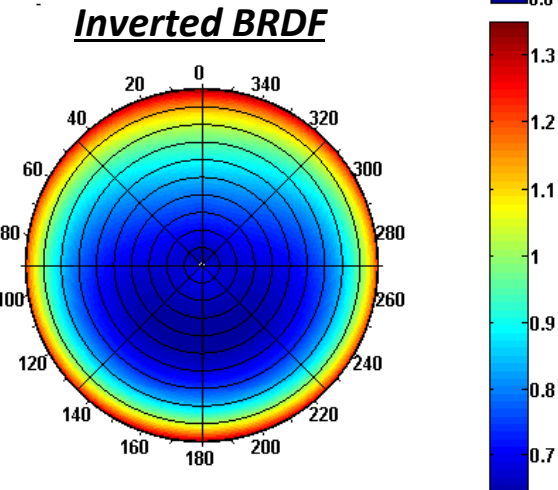
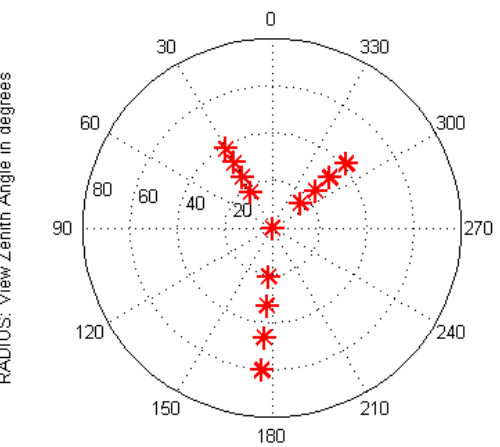


Worst Case
RMS err=0.109
Albedo err=0.0012

The Configuration



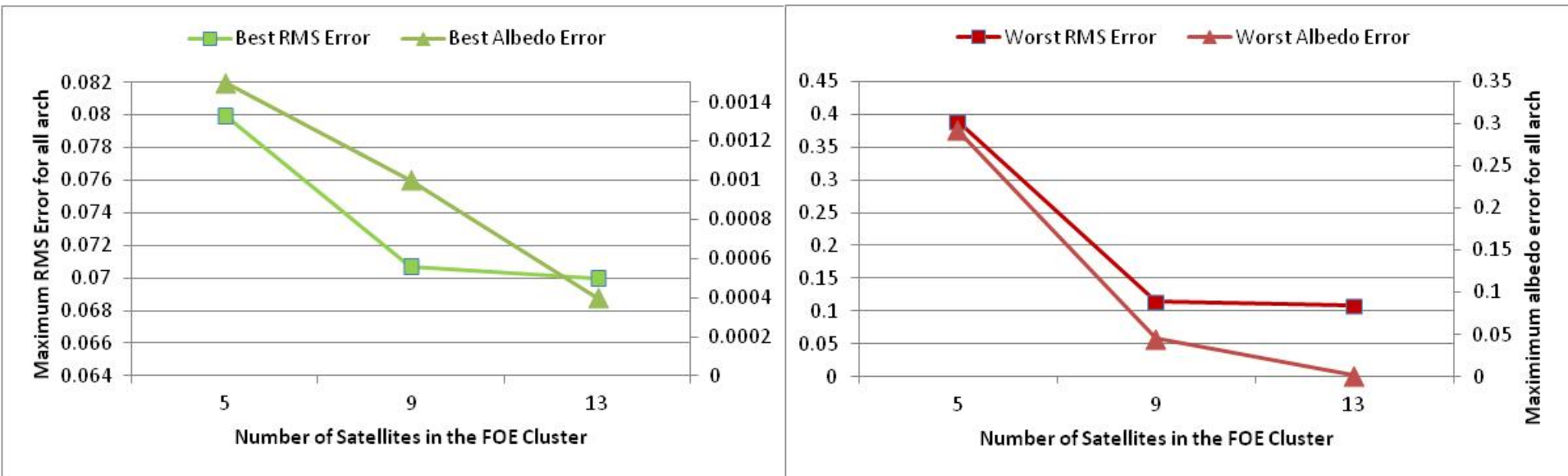
Measurement Angles



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BRDF and Black Sky Albedo errors compared to 'Truth'/CAR Values as a function of #satellites

Important quantification of value to trade against cost of a growing cluster size!

Angular/Area variation for the **BEST FOE geometry**:

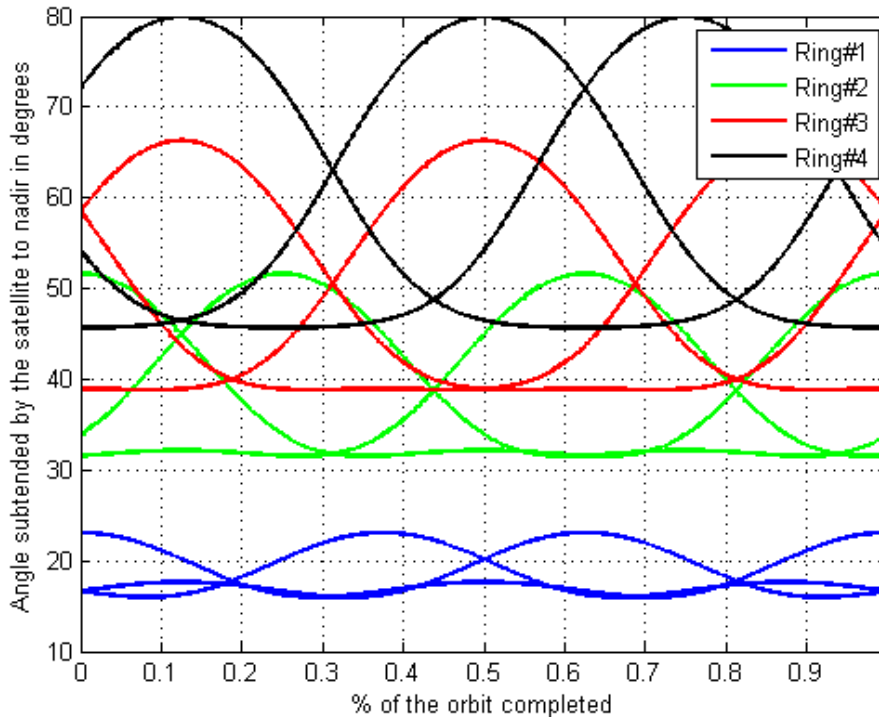
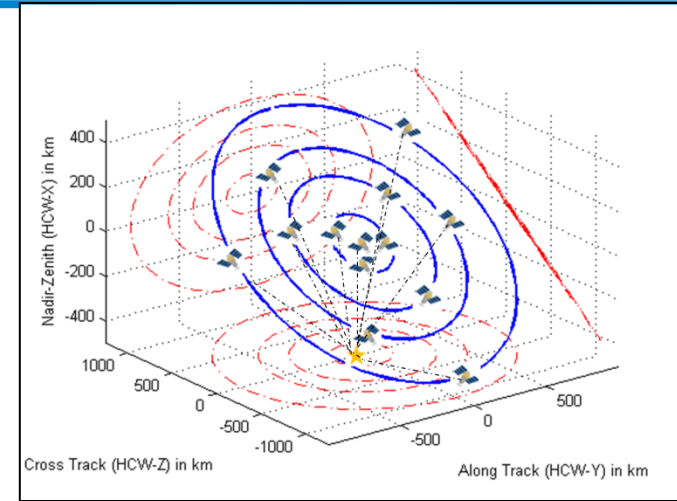
H = 600 km, N = 13 satellites in 4 rings

x0/z0 ratio = 0.4 = 21.8 deg

FOV assumed = 10 deg

Nominal boresight angle = 0, 20, 40, 50, 60 deg

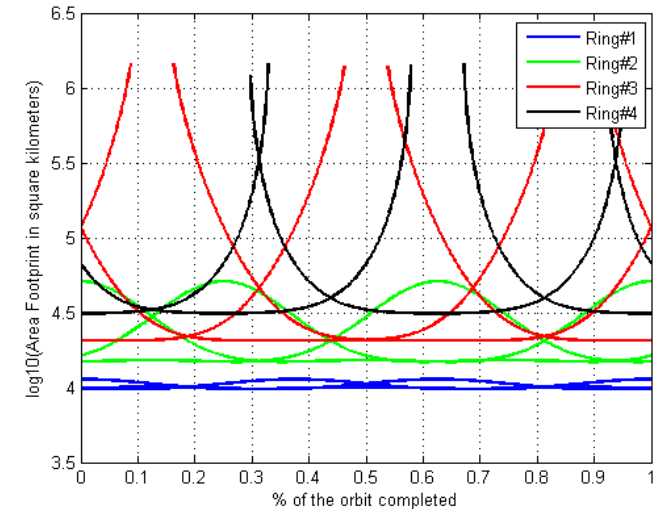
Nominal azimuthal angle = Variation optimized per ring



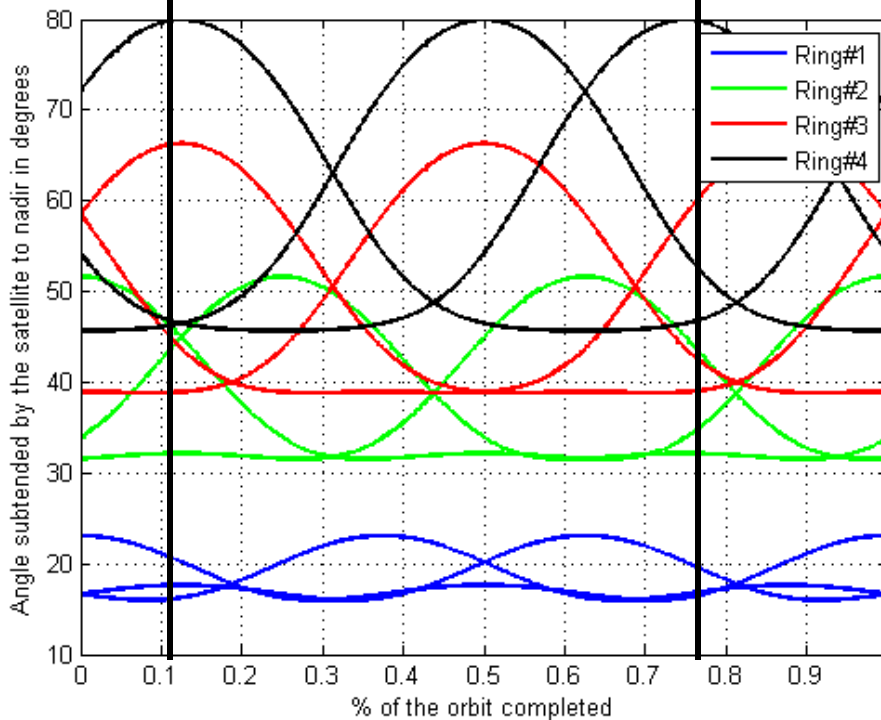
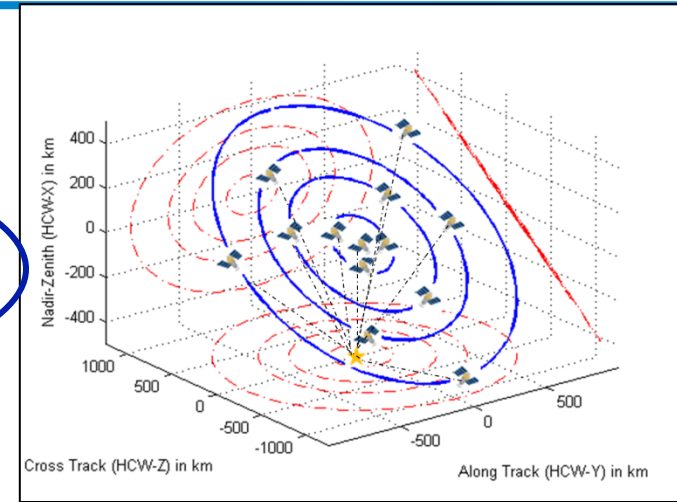
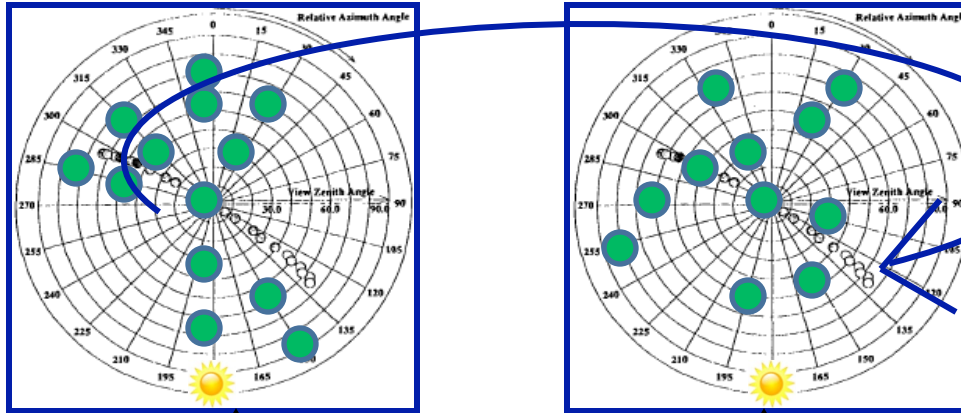
ANGULAR SAMPLING
Required Boresight angle variation:

SPATIAL SAMPLING

Footprint Area variation:

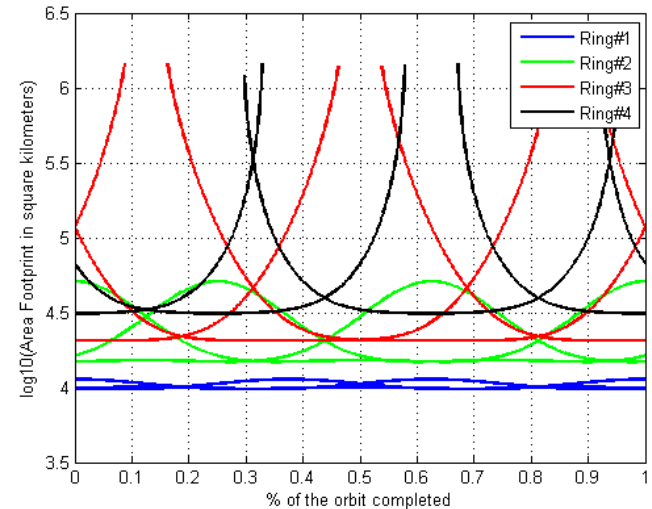


Zenith/Azimuth variation for the **BEST FOE geometry**:



SPATIAL SAMPLING

Footprint Area variation:



ANGULAR SAMPLING
Required Boresight angle variation:

ACDS variation for the **BEST FOE geometry**:

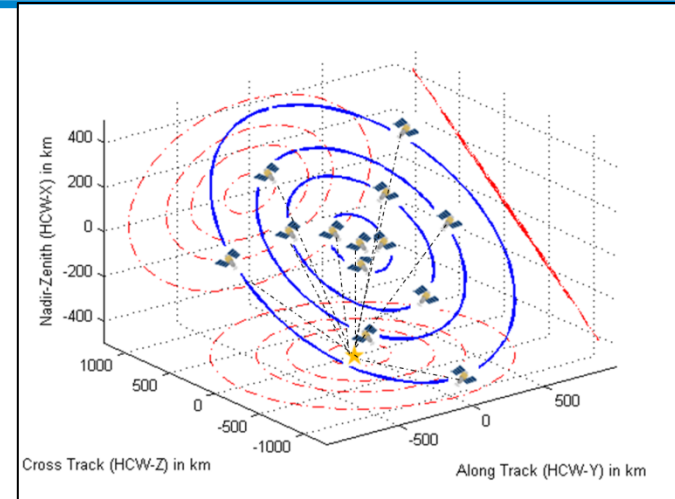
H = 600 km, N = 13 satellites in 4 rings

x_0/z_0 ratio = 0.4 = 21.8 deg

FOV assumed = 10 deg

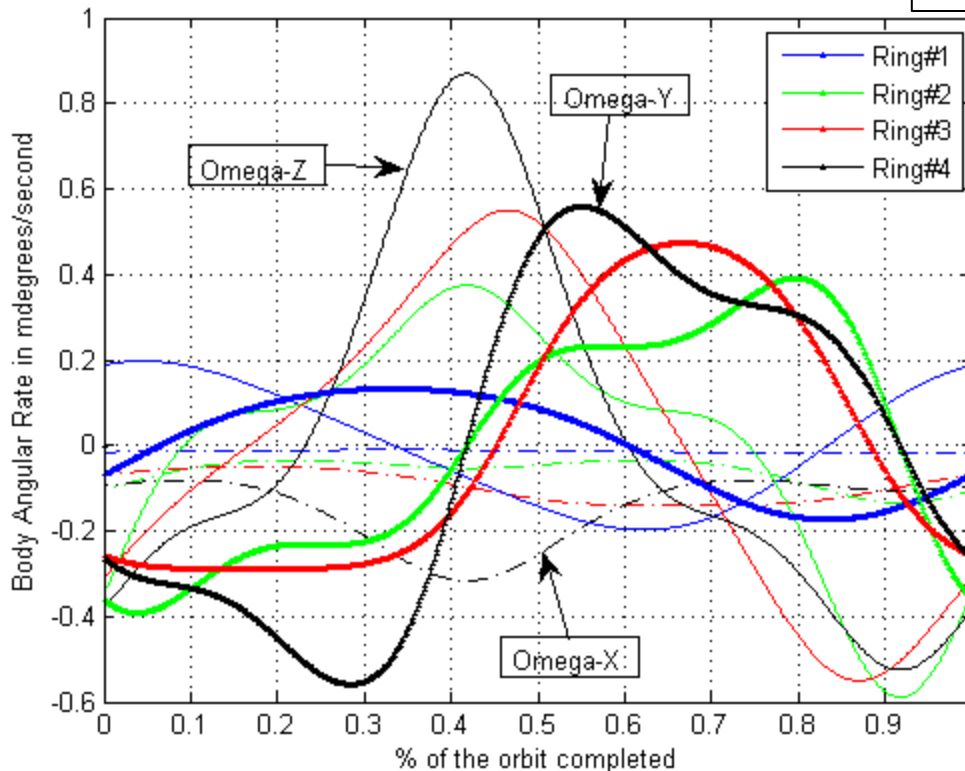
Nominal boresight angle = 0, 20, 40, 50, 60 deg

Nominal azimuthal angle = Variation optimized per ring



Required Slew Rate:

($I = 0.15 \text{ kg-m}^2$
assuming a cubic nanosat)



MAI-400
(Maryland Aerospace) RW



$P \sim 3W$

$m < 0.7 \text{ kg}$

$V \sim 0.5 \text{ cube}$

Max H = 11.8 mNms

Max T = 0.625 mNm

ACDS variation for the **BEST FOE geometry**:

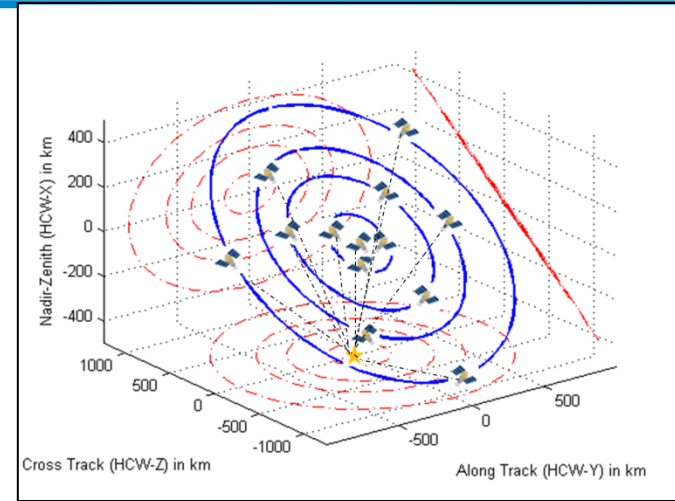
H = 600 km, N = 13 satellites in 4 rings

x0/z0 ratio = 0.4 = 21.8 deg

FOV assumed = 10 deg

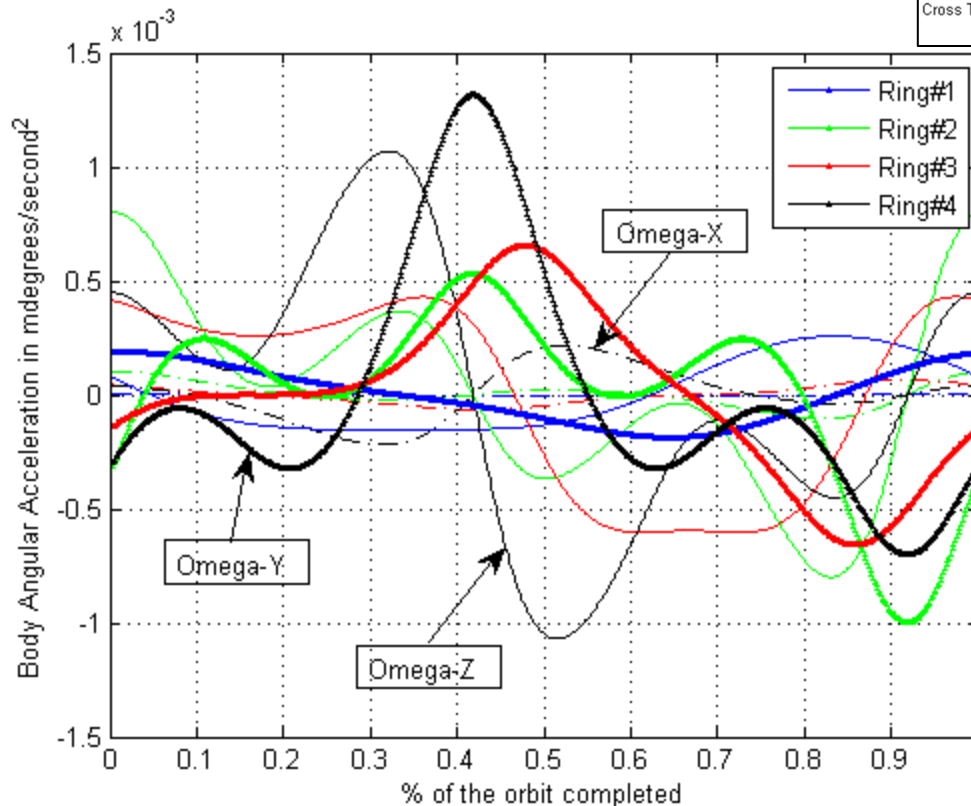
Nominal boresight angle = 0, 20, 40, 50, 60 deg

Nominal azimuthal angle = Variation optimized per ring



Required Angular Acceleration:

(I = 0.15 kg-m²
assuming a cubic nanosat)



**MAI-400
(Maryland Aerospace) RW**



P ~ 3W

m < 0.7 kg

V ~ 0.5 cube

Max H = 11.8

mNm

**Max T = 0.625
mNm**

- **Proposed nanosatellite clusters in formation flight** with VNIR spectrometers to sample the BRDF function as a complement to existing monolithic data products
- Designed a physics-based, **integrated science + systems engineering model** for tradespace exploration to find the “optimal design” for the cluster geometry that will maximize specific BRDF science goals
- Identified **snow albedo** as a critical BRDF application
- Used the tradespaces to **quantify the significance on albedo and BRDF errors** of cluster geometry and orientation, # of Satellites, orbit orientation and azimuthal coverage. Showed that the optimal cluster configuration’s subsystem requirements (e.g. ADCS) is COTS supportable
- **Future Work** includes heuristic optimization of clusters (both modified linearized and global propagation) for albedo accuracy over mission lifetime (cluster dynamics and orbit maintenance). Other critical applications such as GPP will also be realized.

- MIT
 - Prof. Olivier de Weck
 - Prof. Kerri Cahoy
 - Prof. David Miller
- NASA GSFC
 - Charles Gatebe
 - Warren Wiscombe
 - Jacqueline Lemoigne
 - Alexei Lyapustin
 - Ralph Kahn
 - Miguel Roman
 - John Mather Nobel Scholar Award Committee

Questions?